

Manikin Integrated Data Acquisition (MIDAS) Initial Modifications



By

Nabih M. Alem

Aircrew Protection Division

and

James A. Lewis Robert M. Dillard

Research Support Division

19950306 019

December 1994

Approved for public release; distribution unlimited.

United States Army Aeromedical Research Laboratory Fort Rucker, Alabama 36362-0577

Notice

Oualified requesters

Qualified requesters may obtain copies from the Defense Technical Information Center (DTIC), Cameron Station, Alexandria, Virginia 22314. Orders will be expedited if placed through the librarian or other person designated to request documents from DTIC.

Change of address

Organizations receiving reports from the U.S. Army Aeromedical Research Laboratory on automatic mailing lists should confirm correct address when corresponding about laboratory reports.

Disposition

Destroy this document when it is no longer needed. Do not return it to the originator.

Disclaimer

The views, opinions, and/or findings contained in this report are those of the author(s) and should not be construed as an official Department of the Army position, policy, or decision, unless so designated by other official documentation. Citation of trade names in this report does not constitute an official Department of the Army endorsement or approval of the use of such commercial items.

Reviewed:

KEVIN T. MASON

LTC, MC, MFS

Director, Aircrew Protection

ein T Mason

Division

ROGER W. WILLY, O.D., Ph.D.

Chairman, Scientific

Review Committee

Released for publication:

DENNIS F. SHANAHAN Colonel, MC, MFS

Commanding

SECURITY CLAS	SIFICATION O			LDAGE			Form Approved
REPORT DOCUMENTATION							OMB No. 0704-0188
1a. REPORT SE	CURITY CLASS	IFICATION		16. RESTRICTIVE	MARKINGS		
Unclassi				3 DISTRIBUTION	AVAILABILITY OF	REPOR1	
2a. SECURITY CLASSIFICATION AUTHORITY			Approved for public release, distribution				
2b. DECLASSIF	ICATION / DOW	NGRADING SCHEDU		unlimited			·
		ION REPORT NUMBE	R(S)	5. MONITORING	ORGANIZATION RE	PORT N	UMBER(\$)
USAARL	Report No	. 95-11					
6a. NAME OF	PERFORMING	ORGANIZATION	6b. OFFICE SYMBOL (If applicable)	7a. NAME OF MO U.S. Army Me	ONITORING ORGAN	NIZATION	N nd Materiel
U.S. Arm Laborato	-	ical Research	SGRD-UAD-IV	Command (Pro		i cii a	
6c. ADDRESS (d ZIP Code)			y, State, and ZIP C	ode)	
P.O. Box	620577			Fort Detrick Frederick, N		10	
Fort Ruc	ker, AL	36362-0577		Frederick, F	ир 21702-30	12	
8a. NAME OF	ELIMINING / SPC	NSORING	Bb. OFFICE SYMBOL	9. PROCUREMENT	T INSTRUMENT IDE	NTIFICA	TION NUMBER
ORGANIZA		MAJORINAG	(If applicable)				
8c. ADDRESS (City, State, and	i ZIP Code)	•	10. SOURCE OF F	UNDING NUMBER	TASK	WORK UNIT
				ELEMENT NO.	NO.	NO.	ACCESSION NO.
				62787A 30	62787A878	ED	141
12. PERSONAL	AUTHOR(S)		ition system (Mi	Dillard			
13a. TYPE OF Final		13b. TIME C	OVERED TO	14. DATE OF REPO	ORT (Year, Month, December	Day)	15. PAGE COUNT 25
16. SUPPLEME	NTARY NOTA						
10. 3011 22.002			•				
17.	COSATI	CODES	18. SUBJECT TERMS (Continue on reven	se if necessary and	d identif	y by block number)
FIELD	GROUP	SUB-GROUP	prototype, ma	nikin, crash	testing, da	ata	
15	05		acquisition,		hic test de	vice,	
05	09		injury assess				
A prototy Aeromedi manikin, t modificat	pe manikin vical Research the new MID ions to the m	vith internal data as Laboratory. In ad	t-in signal conditioning a description of the	DAS) has been d	olumn and pervi	is repo	by the US Army Hybrid III automotive ort documents the initial rol, communication, and
				The Angertage of	CCHOITY CLASSIFIA	*ATION	
		BILITY OF ABSTRACT			ECURITY CLASSIFIC	LATION	
	SSIFIED/UNLIMI	E INDIVIDUAL	NET. LI DIIL USEKS	22b. TELEPHONE	(Include Area Coo	le) 22c.	OFFICE SYMBOL
Chief.	Science S	Support Center		205-255-6	907	SG	RD-UAX-SI
			Oraniana aditiona 200	- checlate	SECURITY	CLASSI	FICATION OF THIS PAGE

Contents

	Page
List of figures	. 2
Introduction	. 5
Materials and methods	. 6 . 6
Results External software (MIDAS 3.0)	
Discussion	. 9
Summary and conclusions	. 10
References	. 11
Appendix A. Manikin operating procedures	. 12
Appendix B. External manikin software, MIDAS version 3.0	. 14

Accesion For						
NTIS CRA&I VI DTIC TAB II Unannounced II Justification						
By Distribution /						
A	Availability Codes					
Dist Avail and for Special						
A-)						

List of figures

Figure	Pag	е
B-1	Screen printout of the test identification menu which allows the user to enter a test label and additional comment associated with the test	;
B-2.	Screen printout of the data extraction menu where the user selects the source and destination of the data to be extracted	5
B-3.	Screen printout of the data processing and analysis menu. This menu opens several submenus where various types of signal analyses are allowed	5
B-4.	Screen printout of the menu to setup the data acquisition sampling rate and the parameters for impact detection and data extraction from the credit card	5
B-5.	Screen printout of the menu to define channel labels and designate those active channels to be extracted after the test	7
B-6 .	Screen printout of the menu to define the amplifiers gains in the data acquisition system, and to label the engineering units of each active channel.	7
B-7.	Screen printout of the menu to enter the transducers sensitivities of the manikin, as supplied by the manufacturers or determined from calibrations	8
B-8.	Screen printout of the menu to backup credit card data and other labels and settings for later retrieval and analysis	8
B-9.	Screen printout of the manikin controls menu which allows communication with the manikin internal data acquisition sub-system over serial communication port	9
B-10.	Screen printout of the manikin controls menu when the "monitor" mode is activated allowing synchronous serial communication with the manikin	9
B-11.	Screen printout of the menu to setup the colors and communication port parameters. Choosing colors becomes important for laptop LCD screens	0.

List of figures (Continued).

Figure		Page
B-12.	Screen printout of data processing submenu which allows the selection of up to six channels to be graphically displayed, in raw form, on the screen	20
B-13.	Screen printout of the data processing submenu allowing the selection of up to four channels to be plotted in portrait format "A" hard copy for quick look.	21
B-14.	Screen printout of the data processing submenu allowing the selection of up to nine channels to be plotted in portrait format "B" hard copy for quick look	21
B-15.	Screen printout of the data processing sub-menu allowing the selection of up to four channels to be plotted in landscape format "C" hard copy for quick look	22
B-16.	Screen printout of the data processing submenu allowing the selection of up to nine channels to be plotted in landscape format "D" hard copy for quick look	22
B-17.	Screen printout of the data processing submenu allowing the selection of up to eight channels to be plotted in two columns in landscape format "E" hard copy for quick look	23
B-18.	Screen printout of the data processing submenu allowing the selection of up to 18 channels to be plotted in landscape format "F" hard copy for quick look	23
B-19.	Screen printout of the data processing submenu showing test information and allowing the printing of hard copy of pertinent information	24
B-20.	Screen printout of the data processing submenu showing (bottom third) parameters for integral/derivative analysis and allowing the user to modify the parameters	24
B-21.	Screen printout of the data processing submenu showing parameters for over-plot of two signals. This analysis also allows plotting of a single channel	25
B-22.	Screen printout of the data processing submenu showing (bottom third) the parameters selected to analyze a triaxial cluster of transducers	25

This page left blank intentionally.

Introduction

A prototype manikin with internal data acquisition system (MIDAS) has been developed and received by the U.S. Army Aeromedical Research Laboratory (USAARL), Fort Rucker, Alabama, for initial evaluations. New features in this anthropomorphic manikin include a novel design of the spinal column that offers more flexibility in compression and twisting than other manikins, a pelvis that contains the signal conditioning and data acquisition electronics required in most impact tests, and two new upper legs where rechargeable nickel-cadmium cells are placed to provide power supply for the instrumentation and data acquisition system.

As with most prototype systems, MIDAS must be evaluated before it can be put to full use in crash tests and other impact exposures. The evaluation process is simple: expose the manikin to impact tests, collect its biodynamic response signals, then compare the impact response to that of other validated manikins such as the original Hybrid III. This requires that the MIDAS internal electronics and data acquisition system operate properly and allow the data to be extracted in a short time.

Initial bench testing of MIDAS's signal conditioning electronics pointed out areas of potential problems which must be addressed prior to any sled or field testing. First, the battery packs which were installed in the manikin femurs were not supplying adequate power to the electronics. Additionally, the power supply that runs off the standard 120 VAC line and which was delivered with the prototype introduced unacceptable levels of noise to the signal amplifiers. Therefore, a new power supply was needed to replace the current battery packs and provide power from an external source through an umbilical line.

The second area where difficulties were encountered was the software for data acquisition, downloading, and display, which was delivered with the system. Since our immediate goal was the evaluation of the biodynamic response, it was essential that we are able to modify the internal settings of the electronics and to immediately examine the data captured after each test for quick assessment of the manikin performance. Our bench tests also pointed out the need for reliable and flexible software which allows us to achieve these two simple goals: modify the settings and examine the data as quickly as possible.

This report is intended to document the initial improvements made to the MIDAS prototype since its delivery. Because it was only a prototype, modifications were made only to bring the manikin to an acceptable level of reliability during sled tests.

Materials and methods

Electronics hardware

The external power supply was designed to convert 120 VAC to provide several regulated DC voltages to MIDAS's internal computer CPU, memory, disks, transducers, and amplifiers. The box delivered with the manikin had to be reworked by first removing the battery charger subsystem from the power supply box itself. Second, the soldering joints of the cables had to be redone because of potential breakage during anticipated sled mounting of the power supply box.

A serious problem that was identified after extensive debugging was the so-called "common mode problem." This problem caused the signal amplifiers to saturate due to the transducer excitation being tied to system ground. The problem was corrected by floating (relative to system ground) the 15-Volt excitation voltage of each transducer power supply.

Another shortcoming was cumbersome access to the analog signals from the 24 transducers installed in the manikin. Access to these signals is necessary for debugging and maintenance, for bypassing the data acquisition system in case of failure, and for recording the analog signals on an external data recording medium. At one stage during the development and testing of MIDAS, the contractor has built into the wiring access to all the transducer signals. Since this was not part of the contract and because of space constraints, the access cables were removed. In order for USAARL to make its evaluation, it was necessary to reconstruct the access cable and connect it to a breakout box where BNC-type connectors may be used. This proved later to be a valuable modification when it became necessary to record the signals external to the MIDAS hardware.

Data acquisition settings

Precise measurements require accurate knowledge and control of the sampling rate and the gains for each amplifier in the manikin internal data acquisition system. Therefore, it was essential to verify the values of sampling rate and gains claimed by the contractor. We used the voltage insertion method, where a low level sine wave signal, generated by a precision calibrator, was used to replace a transducer and provide an input to the amplifier. By measuring the output of the amplifier, we determined its actual gain as the ratio of its output over the input, and its zero offset as the midpoint of the peak-to-peak range of the output.

Although it was possible to determine the amplifiers' gains by monitoring analog input and output signals, it was necessary to determine the sampling rate of the A/D convertor from the digitized data after it was stored on the thin credit-card memory, then dumped onto a PC disk file and examined. This was done after the first version of the MIDAS software became operational.

External software

As delivered, the operation of MIDAS requires the user to turn on the power supply, modify the amplifier gains and other data acquisition parameters to accommodate specific test conditions, conduct the tests, then remove the credit-card memory module (where the manikin internal computer stored the data) for down-loading and analysis of the digitized data. A new MIDAS program was written to externally control the amplifier settings and sampling rate, and to extract the data from the credit-card and display it for quick look and assessment of the quality of the data acquisition.

Because of its intended use, the MIDAS program was designed to improve the communication by automating most of the functions which were required to be repeated by the user. The program was written in Fortran language (Microsoft Corp, 1988) to be run on an IBM-PC computer under a MS-DOS environment. All test data would be downloaded and archived along with descriptive and calibration data. These archived files may be accessed by MIDAS for further analysis and plotting at a later time after completing a test series. The program produces hard copies of raw and processed signals and summaries of analyzed signals. The program offered significant improvements in the operation of the manikin and, although several additions were contemplated, the first version was deemed adequate for evaluating the results from sled tests.

Results

The bench evaluation of the electronics, wiring, and data acquisition system produced the following observations. Refer to Frisch, Boulay, and Alem, 1994, for a description of the delivered MIDAS and explanation of the components mentioned here.

- 1. The signal conditioner circuitry has no offset capability. Potentiometers on the board that are connected to the sample-and-hold (S/H) amplifiers have no effect on the direct analog or S/H output.
- 2. The voltage substitution calibration, as currently provided, is not useful. The resistor calibration (RCAL) is not implemented.
- 3. Gain and sample rate settings cannot be verified and sometimes are in error. Occasionally the gain sets to gain code 0, and at other times the sample rate sets to lowest value, as evidenced by a very dim data card light-emitting diode (LED).
- 4. For proper operation of the sensors, the power supplies required rewiring to "float" the grounds. The following supplies were affected: +15 volts for all of the accelerometers, +15 volts for all of the load cells, and +10 volts for the angular accelerometer in the head.
 - 5. The specified rates for sampling are only approximate values.

- 6. Sampling at the highest two rates, 5 kilo-Hertz (KHz) and 10 KHz, is not possible with the current system which writes 48 channels to the data card.
- 7. The manikin's internal wiring harness is difficult to check or correct. It should have been of a more replaceable design, so that sensors or cables could be replaced easily. The copper tape shielding is ineffective.
- 8. The definitions of channels 18 and 20 appear to be swapped from their original designations.
- 9. Several signal conditioner hybrid circuits are bad, in one or both channels of the hybrid.
- 10. The L5 load cell channels have noise problems; so do the Head-X and the T1-Y channels.
 - 11. Channel 19 is nonfunctional, producing a 147 Hertz noise signal.
 - 12. Signal conditioner module card #1 has bad U-11 chip.

External software (MIDAS 3.0)

The MIDAS software initially was written to replace the software delivered as part of the contract because that software did not allow sufficient flexibility and reliability in operating the manikin, nor adequate speed in downloading and examining the acquired data. The latest version of MIDAS (version 3.0) was the result of actual tests and experience in the manikin's operation as intensive bench testing was conducted.

The step-by-step procedure to operate the manikin is summarized in Appendix A. This requires the MIDAS software and a PC connected through an RS-232 cable to the manikin internal computer. The capabilities of the MIDAS (version 3.0) software are summarized in Appendix B in the form of a series of screen printouts of the program menus. Although shown in black and white, these "screens" are produced on a VGA monitor in colors which are used for emphasis of key words or important text, and for demarcation between blocks of information. The following features are the highlights of the software.

- 1. The program consists of over 200 subroutines, written in Microsoft Fortran 5 and developed in its entirety at USAARL.
- 2. The program is interactive when necessary and automated when possible, and is entirely menu driven. All of the major submenus in MIDAS are reproduced in Appendix A as screen printouts.

- 3. The program requires an IBM-PC compatible running MS-DOS 5.0 or higher.
- 4. The PC hardware should have a VGA color monitor, although a laptop LCD screen may be used after adjusting the "color" settings to suit the particular screen.
- 5. No mouse is needed but a serial port (COM1 or COM2) must be available to communicate with the manikin internal computer.
- 6. MIDAS software is designed to run independent of the manikin to analyze, plot, and produce hard copies of the analyzed data after all the data from a series of tests have been archived.
- 7. Hard copies are produced in PCL/5 and HP-GL/2 languages and require printers capable of interpreting these languages, such as the HP Laserjet 3 or 4.

Discussion

As with most prototype systems, some adjustments to the delivered version of the system was necessary. The initial modifications were designed primarily to put the manikin in an acceptable and reliable working condition so its performance as a test device could be evaluated. The reliability of the power supply, the calibration of the amplifiers, and the identification and repair (when possible) of known problems were of the utmost concern. Just as critical was the down-load and quick-look capability without which the manikin simply could not have been tested efficiently.

The manikin was designed to be self contained, i.e., capable of operation from an internal power source. The design of the prototype included two sets of nickel-cadmium (nicad) batteries installed in the femur in two cylindrical arrays. Although this approach was discussed and approved during the development phase of the manikin, the execution of this design by a subcontractor resulted in a product that functioned below our expectations. A remedy to this problem was set aside while more important issues were addressed. Thus, the manikin was to operate from external power supply until the internal battery supply issue has been resolved. Indeed, an external power supply was delivered but it also had problems: noise, ground loops, poor shielding, weak soldering connections and unnecessary wiring. All these problem areas were addressed and resolved in the more reliable configuration which was fabricated. This allowed us to advance to the next stage in this research program: the actual sled and controlled drop tests.

Equally important was our ability to program the internal amplifiers' gains and adjust the sampling rate of the manikin data acquisition system to meet the different impact test requirements. The manikin had default settings for all amplifiers and sampling rate which were stored on an erasable/programmable ROM chip in the MIDAS computer and read each time the manikin was turned on. Access to these stored defaults required the "undressing" of the manikin

and the removal of the CPU board, then its installation in a separate but equivalent computer for re-coding. The alternative to this time-consuming task was the transmittal of the desired gain values over the serial communication port every time the manikin was turned on to override the default values.

This approach was an acceptable solution if all goes well. Unfortunately, it took nearly half an hour of interaction between the user and the manikin internal software before the new settings had been transmitted. Frequently, the transmittal was so erratic because of poor COM port hardware and software drivers or because of unexplainable malfunction in the MIDAS internal electronics that it was necessary to reset the internal electronics by turning off the manikin power supply then switching it back on to restart the entire operation. This frustrating and inefficient process was one of the two primary reasons for writing our own external MIDAS control software.

The other motive for rewriting the MIDAS software was our inability to examine the data immediately after it was acquired. This essential requirement was somewhat met by the software as it was delivered. In fact, we were able to quickly remove the credit card and "download" the raw data to a PC file using utilities supplied by the manufacturer of the credit card. Unfortunately, plotting one impact pulse, which may be of 20 or 30 millisecond duration, required the manual plotting of the entire channel signal (1200 msec) in small segments and, when all segments were finally plotted, did not give the user sufficient timing and magnitude information to judge the quality of the test. The tedious process was lengthy and often was abandoned after examining one or two channels inconclusively.

Therefore, the MIDAS program was designed to truly assist the user in making informed decisions regarding a test. Just as the manikin could not have been operated without a reliable power supply, it would have been unwise to operate this new prototype without the benefit of "quick look" and "hard copy" capabilities.

Summary and conclusions

As with most prototypes, initial modifications were necessary to the hardware and the software delivered with the manikin in order to bring it to an acceptable operating status. These include improvements to the external power supply to be used temporarily in lieu of the internal battery system, and a PC-based program (MIDAS) which facilitates the settings of internal amplifiers gains and sampling rate, and allows quick look at the down-loaded data immediately following a test. Eventually, it will be necessary to redesign the internal battery-based power supply so that field tests can be conducted independent of external connections. Also, it will be necessary to modify the internal data acquisition software to improve serial communication and to allow easier modification and retention of default data acquisition parameters.

References

- Frisch, P., Boulay, W., and Alem, N. 1994. <u>Design and development of an enhanced biodynamic manikin</u>. Phase I report for contract DAMD17-90-C-0116. Fort Rucker, AL: U.S. Army Aeromedical Research Laboratory. USAARL contractor report No. CR 94-1.
- Microsoft Corporation. 1989. Fortran reference manual, version 5.0 for MS-DOS operating systems.

Appendix A.

MANIKIN operating procedures.

- 1. Connect RS-232 cable between manikin and PC communications port 1 or 2, as set in the USAARL manikin integrated data acquisition system (MIDAS) program.
 - 2. Install the memory module (credit-card) into the manikin chassis slot.
- 3. Power-up manikin. The green LED on control unit indicates +5 volt power is present. Wait approximately 20 seconds a short flash of the red LED indicates end of boot sequence.
- 4. On PC, run the MIDAS program. Verify sample rate in the "Sampling setup" function, gains and correction factors in "Amplifiers" function, and sensitivities in "Transducers" function.
 - 5. Using "Identify test" function, enter the new test ID and other desired information.
- 6. Use the "Manikin controls" function to send gains and sample rate (press F1). If manikin will not respond, send "escape" characters (press F8) to clear the communications port. If this doesn't help, check for correct communication settings under the "Colors and comm." section, set port at 1200 baud, 8 bits, no parity, and 1 stop bit. Verify cabling is connected to correct port. Manikin uses upper communications port connector on manikin input/output (I/O) card, which is located inside the rib cage.
- 7. "Arm" manikin when ready (press F9). Observe that red LED is on and bright. A dim LED indicates improper sample rate. If LED goes off after 6 seconds (at 2500 Hz sample rate) or after 16 seconds (at 1000 Hz sample rate), then a false trigger has occurred. Turn off manikin power, ensure trigger signal is at ground potential (0 volts), and reapply power to manikin. Resend sample rate and gains after manikin boots up. A steady on LED indicates the proper "armed" condition (data is being written to the card in a circular buffer).
 - 8. Disconnect the RS-232 cable from the manikin.
- 9. Run test. A 0-to-5 volt transition on the trigger signal input will end the 200-points/channel circular buffer and start the post-trigger data recording. When the red LED goes out, the 12,000 post-trigger data points/channel have been written to the card (storage complete).

- 10. Remove data card. Insert data card in the PC, perform "Backup card data" operation, saving the card image to hard disk. Use "Extract signals" to retrieve data from the data card (or from a saved file). Use the "Process" function to view signals.
 - 11. Turn off manikin power if done, or press RESET and go to step 2 for the next test.

Appendix B

External manikin software.

MIDAS version 3.0.

This appendix contains screen printouts of most menus and sub-menus which are available in the MIDAS program. Because they are printed in black and white, much of the color highlights used to emphasize various key words and text, and to separate blocks of information on the screen are, of course, missing. However, these "screens" do provide a printed documentation in lieu of actually running the MIDAS program itself.

Test: LX65-56 ===	MIDAS-3.0 Manikin Integrated DAS 13:01:03
Identify test	
Extract signals	Data ID/Label: LX65-56
Process signals	Description of data:
Sampling setup	Dynamic testing of MIDAS prototype crash manikin
Define channels	Sled acceleration pulse: 10.3 peak G, 23.1 mph
Amplifiers	Seat pitch: +35 degrees
Transducers	Seat pitch: +35 degrees
Backup card data	
Manikin controls	
Colors and comm.	and the second in
Quit to DOS	Identify test data with a unique label to be used in naming binary and exchange data files. Descriptive text may also be added to identify test conditions.

Figure B-1. Screen printout of the test identification menu which allows the user to enter a test label and additional comment associated with the test.

Test: LX65-56	MIDAS-3.0 Manikin Integrated DAS	13:06:32
Identify test	F1 EXTRACT data from THINCARD	20 signals
► Extract signals	into RAM, and store them in MIDAS Internal Format on file	LX65-56.MIF
Process signals	F2 SELECT from listed files => the name and type of MIDAS	LX65-56.MIF LX65-55.MIF
Sampling setup	file to retrieve from disk	LX65-52.MIF LX65-45.MIF
Define channels	F3 RETRIEVE the selected .MIF disk file into RAM memory.	LX65-44.MIF LX65-43.MIF
Amplifiers	disk life inco lair memory.	
Transducers	F4 RESTORE data from 3 backup files now on directory	LX65-56.ini LX65-56.pre
Backup card data	C:\NBDL\BACK\	LX65-56.dat
Manikin controls	F5 EXPORT data to text file:	LX65-56.MXP
Colors and comm.	<u> </u>	
Quit to DOS	Extract data from CC memory and save the desired format. Retrieve binary disk for plotting or export in anoth	files from hard

Figure B-2. Screen printout of the data extraction menu where the user selects the source and destination of the data to be extracted.

MIDAS-3.0	Manikin Integrated DAS	13:06:57
F1	Raw channels	
F2	Test/channel information	
F3	[3] Integral/derivative	
F4	[2] Over-plot signals	
F5	[3] Integral/derivative	
F6	[6] Head triax & HIC	
F7	[5] Triax and resultant	
F8	[5] Triax and resultant	
F10	VGA: 640x480, 80x30	
	Comment might look or hard	dony print
To print. Las	serJet III printer port must a	ilready be
	F2 F3 F4 F5 F6 F7 F8 F10 Plot signals	F1 Raw channels F2 Test/channel information F3 [3] Integral/derivative F4 [2] Over-plot signals F5 [3] Integral/derivative F6 [6] Head triax & HIC F7 [5] Triax and resultant F8 [5] Triax and resultant

Figure B-3. Screen printout of the data processing and analysis menu. This menu opens several submenus where various types of signal analyses are allowed.

Test: LX65-56	= MIDAS-3.0 Manikin Integrated DAS 13:07:29		
Identify test	Sampling rate: 2500.0 Hz/channel (nominal) Actual: 2512.5 Hz per channel		
Extract signals	.398 msec interval		
Process signals	Sampling rate must match rate sent to manikin, other parameters may be re-defined after test.		
▶ Sampling setup	Test duration: 597.0 msec = 1500 pts/ch		
Define channels	Trigger method: SOFTWARE a signal FALLS BELOW		
Amplifiers	threshold threshold		
Transducers	250 pts BEFORE trigger		
Backup card data	Trigger signal: Ch 15 = SLED		
Manikin controls	Threshold: 41 mV = 10.1 G BELOW initial level		
Colors and comm.	Define sampling rate, durations of pre- and post-test		
Quit to DOS	calibration lengths, pre-trigger delay and duration of test, all depending on the size of CC memory.		

Figure B-4. Screen printout of the menu to setup the data acquisition sampling rate and the parameters for impact detection and data extraction from the credit card.

	Test: LX65-56 ===	MIDAS-3.0	Manikin Integrated DAS	13:07:56	
	Identify test	(20) ON Label	Channel description	A/D channel	
	Extract signals	= ====================================	Tood V (forward) aggal	01	
	Process signals	01 AX-HEAD 02 AY-HEAD 03 AZ-HEAD	Head Y (lateral) accel	03 05	
	Sampling setup	04 AAY-HEA 05 FZ-NECK	D Head Y (pitch) accel	07 09	
•	Define channels	06 FX-NECK 07 MY-NECK	Neck X (forward) force	11 13	
	Amplifiers	08 AX-T1 09 AY-T1	T1 X (forward) accel	15 17	
	Transducers	10 AZ-T1 11 FZ-T1	T1 Z (vertical) accel T1 Z (vertical) force	19 21	
	Backup card data	12 FX-T1	T1 X (forward) force T1 Y (pitch) moment	23 25	
	Manikin controls	13 MY-T1 14 MX-T1 15 SLED	T1 X (roll) moment Sled (external) accel	27 29	
	Colors and comm.			eering units	
	Quit to DOS Define and describe channels. Name engineering units for the signals. Designate signals which are to be extracted, stored, and displayed.				

Figure B-5. Screen printout of the menu to define channel labels and designate those active channels to be extracted after the test.

Test: LX65-56	mIDAS-3.0	Maniki	n Integrated	DAS	13:09:21
Identify test	(20) Ch Label	DASS Gain	Correction Exp/actual		Units
Extract signals	01 AX-HEAD	100	1.000	.0000	= =====
Process signals	02 AY-HEAD 03 AZ-HEAD	25 100	1.087	.0000	G G
Sampling setup	04 AAY-HEAD	100	1.000	.0000	krad/s2 kN
Define channels	05 FZ-NECK 06 FX-NECK	500 250	1.000 1.000	.0000	kN
► Amplifiers	07 MY-NECK 08 AX-T1	250 25	1.000 1.000	.0000 .0000	kN-m G
_	09 AY-T1 10 AZ-T1	20 25	1.000	.0000	G G
Transducers	11 FZ-T1	1000	1.000	.0000	kN kN
Backup card data	12 FX-T1 13 MY-T1	250 250	1.000 1.000	.0000 .0000	kN-m
Manikin controls	14 MX-T1 15 SLED	250 1	1.000 .1000	.0000	kN-m G
Colors and comm.	Amplifier gai				nding them
Quit to DOS	to the DASS i intercepts of	n the π	anikin. Als	o, define	slopes and

Figure B-6. Screen printout of the menu to define the amplifiers gains in the data acquisition system, and to label the engineering units of each active channel.

_	Test: LX65-56 ===	MI	DAS-3.0	Manikin Integra	ted DAS	13:09:40
	Identify test	(20) Ch	Label	mV / Unit (@ gain=1)	Units/ADC (overall)	Units
	Extract signals	01	AX-HEAD	.5050		G
	Process signals	02 03	AY-HEAD AZ-HEAD	.5380 .5570	.9076E-01 .2192E-01	G G
	Sampling setup	04	AAY-HEAD FZ-NECK		.3083E-02	krad/s2 kN
	Define channels	06	FX-NECK	2.800	.1744E-02 .5679E-04	kN kN-m
	Amplifiers	07 08 09	MY-NECK AX-T1	85.98 2.430	.2009E-01	G
 	► Transducers		AY-T1 AZ-T1	2.440	.2501E-01 .2001E-01	G G
	Backup card data	11 12	FZ-T1 FX-T1	.8913 2.511	.1370E-02 .1945E-02	kN kN
	Manikin controls	13 14	MY-T1 MX-T1	47.60 46.67	.1026E-03 .1046E-03	kN-m kN-m
	Colors and comm.	15	SLED	4.070	.2999	G
			sducers s	ensitivities (e. rs to convert A/	g., mV/G) may	be edited.
	Quit to DOS	unit	s are sho	wn here for veri	fication.	

Figure B-7. Screen printout of the menu to enter the transducers sensitivities of the manikin, as supplied by the manufacturers or determined from calibrations.

Test: LX65-56 ===	MIDAS-3.0 Manikin Integrated DAS 13:10:43							
Identify test	Page Save credit card data on backup files							
Extract signals								
Process signals	Backup subdirectory: C:\NBDL\BACK\							
Sampling setup	Configuration file: 1.INI							
Define channels	Pre-trigger data: 2.PRE							
Amplifiers	Post-trigger data: 3.DAT							
Transducers	Esc Do NOT backup, CANCEL edit changes							
▶ Backup card data	F1 Do NOT backup, but SAVE file names							
Manikin controls	F9 RE-WRITE modified file: LX65-56.MIF							
Colors and comm.								
Quit to DOS	pre- & post-trigger (.PRE, .DAT) data files. Manikin configuration will also be saved on (.CFG) file.							

Figure B-8. Screen printout of the menu to backup credit card data and other labels and settings for later retrieval and analysis.

Test: LX65-56 ===	MIDAS-3.0 Manikin Integrated DAS 13:11:03 ==						
Identify test	F1 send SAMPLING rate and ALL GAINS to DASS						
Extract signals	F2 send only SAMPLING rate = 2512 Hz to DASS						
Process signals	F3 send GAIN = $100x$ <7> to Ch 01 = AX-HEAD						
Sampling setup	F4 request pre-test calibration in DASS						
Define channels	F5 MONITOR: terminal mode over COM2 serial port						
Amplifiers	F6 request post-test calibration in DASS						
Transducers	F7 future use						
Backup card data	F8 send 3 ESC to DASS (to clear hung COM2 port)						
▶ Manikin controls	F9 arm manikin (put DASS in acquisition mode)						
Colors and comm.							
	Control Panel for manikin: initializes manikin to						
Quit to DOS	specified configuration, arms manikin for trigger detection, and downloads acquired data from card.						

Figure B-9. Screen printout of the manikin controls menu which allows communication with the manikin internal data acquisition subsystem over serial communication port.

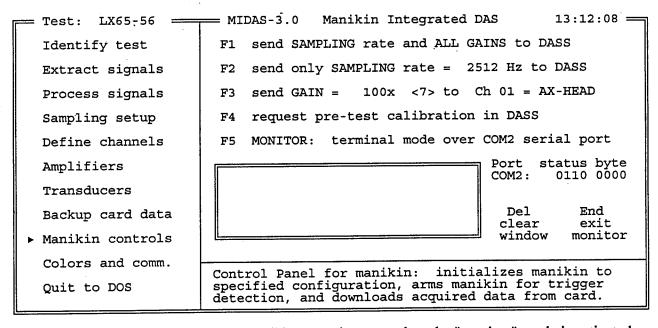


Figure B-10. Screen printout of the manikin controls menu when the "monitor" mode is activated allowing synchronous serial communication with the manikin.

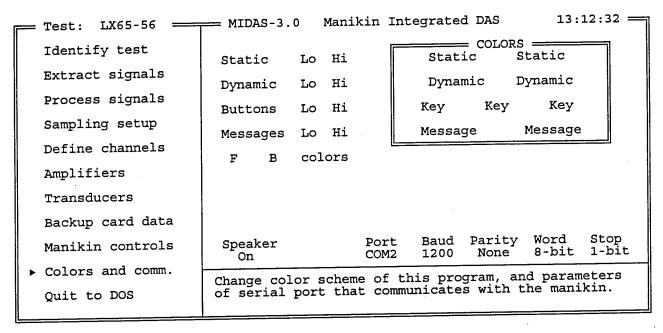


Figure B-11. Screen printout of the menu to setup the colors and communication port parameters. Choosing colors becomes important for laptop LCD screens.

			77.0	Test/channel information
F1	Raw	channels	F2	
F3	[1]	Cross-plot signals	F4	[2] Over-plot signals
F5	[3]	Integral/derivative	F6	[4] 1st/2nd integrals
F7	[5]	Triax and resultant	F8	[6] Head triax & HIC
F9	Exe	cute all ◆▶ functions	F10	VGA: 640x480, 80x30
Fn or	Enter	screen output Alt -	Fn parar	meters Tab form/graph Esc exit
01 = AX-HE 02 = AY-HE 03 = AZ-HE 04 = AAY-H 05 = FZ-NE 06 = FX-NE 07 = MY-NE 08 = AX-TI 09 = AY-TI 10 = AZ-TI	EAD = EAD = HEAD = ECK = ECK = L = L = L		01 02 03 05 06 15 Analysis	span GAPRTST3

Figure B-12. Screen printout of data processing sub-menu which allows the selection of up to six channels to be graphically displayed, in raw form, on the screen.

F1	Raw channels	F2 Test/channel information
F3	[1] Cross-plot signals	F4 [2] Over-plot signals
F5	[3] Integral/derivative	F6 [4] 1st/2nd integrals
F7	[5] Triax and resultant	F8 [6] Head triax & HIC
F9	Execute all ◆ functions	F10 VGA: 640x480, 80x30
	Alt - Fn	parameters Tab form/graph
Ctrl - Fr	n hardcopy output	Esc exit
01 = AX-HEA 02 = AY-HEA 03 = AZ-HEA 04 = AAY-HEA 05 = FZ-NEC 06 = FX-NEC 07 = MY-NEC 08 = AX-T1	AD = Head Y (lateral) accel AD = Head Z (vertical) accel EAD = Head Y (pitch) accel CK = Neck Z (vertical) force CK = Neck X (forward) force CK = Neck Y (pitch) moment	01 02 03
	<pre>= T1 Y (lateral) accel = T1 Z (vertical) accel</pre>	HARDCOPY 6APRTST3 ↑A

Figure B-13. Screen printout of the data processing submenu allowing the selection of up to four channels to be plotted in portrait format "A" hard copy for quick look.

F1	Raw	channels	F2	Test/channel information
F3	[1]	Cross-plot signals	F4	[2] Over-plot signals
F 5	[3]	Integral/derivative	F6	[4] 1st/2nd integrals
F7	[5]	Triax and resultant	F8	[6] Head triax & HIC
F9	Exe	cute all ◆▶ functions	F10	VGA: 640x480, 80x30
Ctrl - Fn	ı ha:	Alt - Fn	param	neters Tab form/graph Esc exit
01 = AX-HEA 02 = AY-HEA 03 = AZ-HEA 04 = AAY-HE 05 = FZ-NEC 06 = FX-NEC 07 = MY-NEC 08 = AX-T1 09 = AY-T1 10 = AZ-T1	D = 1D =	Head X (forward) accel Head Y (lateral) accel Head Z (vertical) accel Head Y (pitch) accel Neck Z (vertical) force Neck X (forward) force Neck Y (pitch) moment T1 X (forward) accel T1 Y (lateral) accel T1 Z (vertical) accel	01 - 02 - 03 - 04 - 05 - 06 - 07 - 08 - 09 - HARD	DCOPY 6APRTST3 2B

Figure B-14. Screen printout of the data processing submenu allowing the selection of up to nine channels to be plotted in portrait format "B" hard copy for quick look.

F1	Raw	channels	F2	Test/channel information
F3	[1]	Cross-plot signals	F4	[2] Over-plot signals
F5	[3]	Integral/derivative	F6	[4] 1st/2nd integrals
F7	[5]	Triax and resultant	F8	[6] Head triax & HIC
F9	Exe	cute all ◆ functions	F10	VGA: 640x480, 80x30
		Alt -	Fn param	eters Tab form/graph
Ctrl - F	n ha:	rdcopy output		Esc exit
01 = AX-HE 02 = AY-HE 03 = AZ-HE 04 = AAY-H 05 = FZ-NE 06 = FX-NE 07 = MY-NE	AD = AD = EAD = ECK = EC	Head X (forward) accel Head Y (lateral) accel Head Z (vertical) accel Head Y (pitch) accel Neck Z (vertical) force Neck X (forward) force Neck Y (pitch) moment	03	
08 = AX-T1 09 = AY-T1 10 = AZ-T1	. = . =	T1 X (forward) accel	04 HARDCOPY	6APRTST3 ↑C

Figure B-15. Screen printout of the data processing submenu allowing the selection of up to four channels to be plotted in landscape format "C" hard copy for quick look.

F1.	Raw	channels	F2	Test/channel information
F3	[1]	Cross-plot signals	F4	[2] Over-plot signals
F 5	[3]	Integral/derivative	F6	[4] 1st/2nd integrals
F7	[5]	Triax and resultant	F8	[6] Head triax & HIC
F9	Exe	cute all ◆ functions	F10	VGA: 640x480, 80x30
		Alt - Fr	param	eters Tab form/graph
Ctrl - Fn	ha:	rdcopy output		Esc exit
01 = AX-HEA 02 = AY-HEA 03 = AZ-HEA 04 = AAY-HE 05 = FZ-NEC 06 = FX-NEC 07 = MY-NEC 08 = AX-T1 09 = AY-T1 10 = AZ-T1	AD = AD = EAD = EK = E	Head X (forward) accel 01 Head Y (lateral) accel 02 Head Z (vertical) accel 03 Head Y (pitch) accel 04 Neck Z (vertical) force 05 Neck X (forward) force 06 Neck Y (pitch) moment 07 T1 X (forward) accel 05 T1 Y (lateral) accel 05 T1 Z (vertical) accel HA		6APRTST3 2D

Figure B-16. Screen printout of the data processing submenu allowing the selection of up to nine channels to be plotted in landscape format "D" hard copy for quick look.

F1	Raw channels	F2 Test/channel information
F3	[1] Cross-plot signals	F4 [2] Over-plot signals
F5	[3] Integral/derivative	F6 [4] 1st/2nd integrals
F7	[5] Triax and resultant	F8 [6] Head triax & HIC
F 9	Execute all ◆ functions	F10 VGA: 640x480, 80x30
	Alt -	Fn parameters Tab form/graph
Ctrl - F	n hardcopy output	Esc exit
01 = AX-HE 02 = AY-HE	AD = Head Y (lateral) accel	01
$\begin{array}{ccc} 03 &=& AZ-HE \\ 04 &=& AAY-E \end{array}$	EAD = Head Y (pitch) accel	02
05 = FZ-NE 06 = FX-NE	CK = Neck Z (vertical) force CK = Neck X (forward) force	03
07 = MY - NF $08 = AX - TI$	= T1 X (forward) accel	04
09 = AY-TI 10 = AZ-TI	/	HARDCOPY 6APRTST3 3E

Figure B-17. Screen printout of the data processing submenu allowing the selection of up to eight channels to be plotted in two columns in landscape format "E" hard copy for quick look.

F1	Raw	channels	F2	Test/channel information
F3	[1]	Cross-plot signals	F4	[2] Over-plot signals
F5	[3]	Integral/derivative	F6	[4] 1st/2nd integrals
F7	[5]	Triax and resultant	F8	[6] Head triax & HIC
F9	Exec	cute all <> functions	F10	VGA: 640x480, 80x30
Ctrl - F	n har	Alt -	Fn param	eters Tab form/graph Esc exit
01 = AX-HEZ 02 = AY-HEZ 03 = AZ-HEZ 04 = AAY-HI 05 = FZ-NEZ 06 = FX-NEZ 07 = MY-NEZ 08 = AX-T1 09 = AY-T1 10 = AZ-T1	AD = AD = EAD = CK = CK = CK = CK =	Head X (forward) accel Head Y (lateral) accel Head Z (vertical) accel Head Y (pitch) accel Neck Z (vertical) force Neck X (forward) force Neck Y (pitch) moment T1 X (forward) accel T1 Y (lateral) accel T1 Z (vertical) accel	01 02 03 04 05 06 08 09 HARDCOPY	10

Figure B-18. Screen printout of the data processing sub-menu allowing the selection of up to 18 channels to be plotted in landscape format "F" hard copy for quick look.

F1	Raw channels	4 >	F2	Test/channel information
F3	[3] Integral/derivative		F4	[2] Over-plot signals
F5	[3] Integral/derivative		F6	[2] Over-plot signals
F7	[1] Cross-plot signals		F8	[2] Over-plot signals
F9	Execute all ◆ functions		F10	VGA: 640x480, 80x30
	Test LX65-56 1500 pt	s @ 25	512.5 H	z = 597.0 ms long
	1500 ==		for har	
	Descript.: Dynamic testing of Sled acceleration Seat pitch: +35	purse:	10.3	type crash manikin peak G, 23.1 mph
	20 channels: 01 - 20	nalysis Spar		9.7 ms .0 to 59.7 ms.

Figure B-19. Screen printout of the data processing submenu showing test information and allowing the printing of hard copy of pertinent information.

	F1	Raw	channels		•	F2	Test	/channel	information
	F3	[3]	Integral/derivati	ve		F4	[2]	Over-plo	t signals
4>	F5	[3]	Integral/derivati	ve		F6	[2]	Over-plo	t signals
	F7	[1]	Cross-plot signal	s		F8	[2]	Over-plo	t signals
	F9	Exec	cute all 🔸 functi	ons		F10	VGA:	640x48	0, 80x30
Fı	n or	Enter	screen output	Alt -	- Fn	para	ameters	Tab	form/graph
Ct	trl - F	n hai	cdcopy output	Ins	mark	for	hardcop	У	Esc exit
Sig	gnals:	05 F2	Z-NECK	First times	integ	gral		First d times	erivative 1.0000
	ilter: (LP)	100 24	Hz dB/oct Set	integ	gral E	BEG.	= 0		
Lá	abels:	Head (kN)	axial (Z) force					Head Z (G)	
	65-56 otion:		run, WITHOUT cush	ion					

Figure B-20. Screen printout of the data processing submenu showing (bottom third) parameters for integral/derivative analysis and allowing the user to modify the parameters.

r			
F1	Raw channels		F2 Test/channel information
F3	[3] Integral/derivative	∢ ►	F4 [2] Over-plot signals
F5	[3] Integral/derivative		F6 [2] Over-plot signals
F7	[1] Cross-plot signals		F8 [2] Over-plot signals
F9	Execute all ◆ functions		F10 VGA: 640x480, 80x30
Fn or	Enter screen output Alt	– Fn	parameters Tab form/graph
Ctrl - F	n hardcopy output Ins	mark	for hardcopy Esc exit
Signals:	10 AZ-T1 (Left vertical)	00 Seat pit (Right vertical)
Filter: (LP)	100 Hz 24 dB/oct		
Labels:	Neck (C1) axial force (kN)		
LX65-56 Caption:	7 Gx run WITHOUT cushion		

Figure B-21. Screen printout of the data processing submenu showing parameters for over-plot of two signals. This analysis also allows plotting of a single channel.

F1	Raw channels	<u>-</u>	F2	Test/channel information	L
F3	[3] Integral/de	Integral/derivative		[2] Over-plot signals	
F5	[3] Integral/de	Integral/derivative		[2] Over-plot signals	
∢⊳ F7	[5] Triax and re	esultant	F8	[2] Over-plot signals	
F9	Execute all <>	functions	F10	VGA: 640x480, 80x30	
Fn or Enter screen output Alt - Fn parameters Tab form/graph					
Ctrl — Fn hardcopy output Ins mark for hardcopy Esc exit					
Signals:	01 AX-HEAD 02	AY-HEAD	03 AZ-HEAD	Tri-axial resultant	
Filter: (LP)	100 Hz 24 dB/oct				
Labels:	Head X Head (G) (G	ad Y	Head Z (G)	Head resultant accel (G)	
LX65-56 Caption:	7 Gx run WITHOUT	cushion			

Figure B-22. Screen printout of the data processing submenu showing (bottom third) the parameters selected to analyze a triaxial cluster of transducers.

Initial distribution

Commander, U.S. Army Natick Research, Development and Engineering Center ATTN: SATNC-MIL (Documents Librarian) Natick, MA 01760-5040

Chairman
National Transportation Safety Board
800 Independence Avenue, S.W.
Washington, DC 20594

Commander
10th Medical Laboratory
ATTN: Audiologist
APO New York 09180

Naval Air Development Center Technical Information Division Technical Support Detachment Warminster, PA 18974

Commanding Officer, Naval Medical Research and Development Command National Naval Medical Center Bethesda, MD 20814-5044

Deputy Director, Defense Research and Engineering ATTN: Military Assistant for Medical and Life Sciences Washington, DC 20301-3080

Commander, U.S. Army Research Institute of Environmental Medicine Natick, MA 01760

Library Naval Submarine Medical Research Lab Box 900, Naval Sub Base Groton, CT 06349-5900 Executive Director, U.S. Army Human Research and Engineering Directorate ATTN: Technical Library Aberdeen Proving Ground, MD 21005

Commander Man-Machine Integration System Code 602 Naval Air Development Center Warminster, PA 18974

Commander Naval Air Development Center ATTN: Code 602-B Warminster, PA 18974

Commanding Officer Armstrong Laboratory Wright-Patterson Air Force Base, OH 45433-6573

Director Army Audiology and Speech Center Walter Reed Army Medical Center Washington, DC 20307-5001

Commander/Director
U.S. Army Combat Surveillance
and Target Acquisition Lab
ATTN: SFAE-IEW-JS
Fort Monmouth, NJ 07703-5305

Director
Federal Aviation Administration
FAA Technical Center
Atlantic City, NJ 08405

Director Walter Reed Army Institute of Research Washington, DC 20307-5100 Commander, U.S. Army Test and Evaluation Command Directorate for Test and Evaluation ATTN: AMSTE-TA-M (Human Factors Group) Aberdeen Proving Ground, MD 21005-5055

Naval Air Systems Command Technical Air Library 950D Room 278, Jefferson Plaza II Department of the Navy Washington, DC 20361

Director
U.S. Army Ballistic
Research Laboratory
ATTN: DRXBR-OD-ST Tech Reports
Aberdeen Proving Ground, MD 21005

Commander
U.S. Army Medical Research
Institute of Chemical Defense
ATTN: SGRD-UV-AO
Aberdeen Proving Ground,
MD 21010-5425

Commander
USAMRMC
ATTN: SGRD-RMS
Fort Detrick, Frederick, MD 21702-5012

HQ DA (DASG-PSP-O) 5109 Leesburg Pike Falls Church, VA 22041-3258

Harry Diamond Laboratories ATTN: Technical Information Branch 2800 Powder Mill Road Adelphi, MD 20783-1197 U.S. Army Materiel Systems
Analysis Agency
ATTN: AMXSY-PA (Reports Processing)
Aberdeen Proving Ground
MD 21005-5071

U.S. Army Ordnance Center and School LibrarySimpson Hall, Building 3071Aberdeen Proving Ground, MD 21005

U.S. Army Environmental
Hygiene Agency
ATTN: HSHB-MO-A
Aberdeen Proving Ground, MD 21010

Technical Library Chemical Research and Development Center Aberdeen Proving Ground, MD 21010-5423

Commander
U.S. Army Medical Research
Institute of Infectious Disease
ATTN: SGRD-UIZ-C
Fort Detrick, Frederick, MD 21702

Director, Biological Sciences Division Office of Naval Research 600 North Quincy Street Arlington, VA 22217

Commandant
U.S. Army Aviation
Logistics School ATTN: ATSQ-TDN
Fort Eustis, VA 23604

Headquarters (ATMD)
U.S. Army Training
and Doctrine Command
ATTN: ATBO-M
Fort Monroe, VA 23651

IAF Liaison Officer for Safety USAF Safety Agency/SEFF 9750 Avenue G, SE Kirtland Air Force Base NM 87117-5671

Naval Aerospace Medical Institute Library Building 1953, Code 03L Pensacola, FL 32508-5600

Command Surgeon
HQ USCENTCOM (CCSG)
U.S. Central Command
MacDill Air Force Base, FL 33608

Director
Directorate of Combat Developments
ATTN: ATZQ-CD
Building 515
Fort Rucker, AL 36362

U.S. Air Force Institute of Technology (AFIT/LDEE) Building 640, Area B Wright-Patterson Air Force Base, OH 45433

Henry L. Taylor Director, Institute of Aviation University of Illinois-Willard Airport Savoy, IL 61874

Chief, National Guard Bureau ATTN: NGB-ARS Arlington Hall Station 111 South George Mason Drive Arlington, VA 22204-1382

AAMRL/HEX Wright-Patterson Air Force Base, OH 45433 Commander
U.S. Army Aviation and Troop Command
ATTN: AMSAT-R-ES
4300 Goodfellow Bouvelard
St. Louis, MO 63120-1798

U.S. Army Aviation and Troop Command Library and Information Center Branch ATTN: AMSAV-DIL4300 Goodfellow BoulevardSt. Louis, MO 63120

Federal Aviation Administration Civil Aeromedical Institute Library AAM-400A P.O. Box 25082 Oklahoma City, OK 73125

Commander
U.S. Army Medical Department
and School
ATTN: Library
Fort Sam Houston, TX 78234

Commander
U.S. Army Institute of Surgical Research
ATTN: SGRD-USM
Fort Sam Houston, TX 78234-6200

Air University Library (AUL/LSE)
Maxwell Air Force Base, AL 36112

Product Manager Aviation Life Support Equipment ATTN: SFAE-AV-LSE 4300 Goodfellow Boulevard St. Louis, MO 63120-1798 Commander and Director
USAE Waterways Experiment Station
ATTN: CEWES-IM-MI-R,
CD Department
3909 Halls Ferry Road
Vicksburg, MS 39180-6199

Commanding Officer Naval Biodynamics Laboratory P.O. Box 24907 New Orleans, LA 70189-0407

Assistant Commandant U.S. Army Field Artillery School ATTN: Morris Swott Technical Library Fort Sill, OK 73503-0312

Mr. Peter Seib Human Engineering Crew Station Box 266 Westland Helicopters Limited Yeovil, Somerset BA20 2YB UK

U.S. Army Dugway Proving Ground Technical Library, Building 5330 Dugway, UT 84022

U.S. Army Yuma Proving Ground Technical Library Yuma, AZ 85364

AFFTC Technical Library 6510 TW/TSTL Edwards Air Force Base, CA 93523-5000

Commander Code 3431 Naval Weapons Center China Lake, CA 93555 Aeromechanics Laboratory U.S. Army Research and Technical Labs Ames Research Center, M/S 215-1 Moffett Field, CA 94035

Sixth U.S. Army ATTN: SMA Presidio of San Francisco, CA 94129

Commander U.S. Army Aeromedical Center Fort Rucker, AL 36362

Strughold Aeromedical Library Document Service Section 2511 Kennedy Circle Brooks Air Force Base, TX 78235-5122

Dr. Diane Damos Department of Human Factors ISSM, USC Los Angeles, CA 90089-0021

U.S. Army White Sands
Missile Range
ATTN: STEWS-IM-ST
White Sands Missile Range, NM 88002

Director, Airworthiness Qualification Test Directorate (ATTC) ATTN: STEAT-AQ-O-TR (Tech Lib) 75 North Flightline Road Edwards Air Force Base, CA 93523-6100

Ms. Sandra G. Hart Ames Research Center MS 262-3 Moffett Field, CA 94035

Commander
USAMRMC
ATTN: SGRD-UMZ
Fort Detrick, Frederick, MD 21702-5009

Commander
U.S. Army Health Services Command
ATTN: HSOP-SO
Fort Sam Houston, TX 78234-6000

U. S. Army Research Institute Aviation R&D Activity ATTN: PERI-IR Fort Rucker, AL 36362

Commander U.S. Army Safety Center Fort Rucker, AL 36362

U.S. Army Aircraft Development Test Activity ATTN: STEBG-MP-P Cairns Army Air Field Fort Rucker, AL 36362

Commander
USAMRMC
ATTN: SGRD-PLC (COL R. Gifford)
Fort Detrick, Frederick, MD 21702

TRADOC Aviation LO Unit 21551, Box A-209-A APO AE 09777

Netherlands Army Liaison Office Building 602 Fort Rucker, AL 36362

British Army Liaison Office Building 602 Fort Rucker, AL 36362

Italian Army Liaison Office Building 602 Fort Rucker, AL 36362 Directorate of Training Development Building 502 Fort Rucker, AL 36362

Chief USAHEL/USAAVNC Field Office P. O. Box 716 Fort Rucker, AL 36362-5349

Commander, U.S. Army Aviation Center and Fort Rucker ATTN: ATZQ-CG Fort Rucker, AL 36362

Dr. Sehchang HahDept. of Behavior Sciences and Leadership, Building 601, Room 281U. S. Military AcademyWest Point, NY 10996-1784

Canadian Army Liaison Office Building 602 Fort Rucker, AL 36362

German Army Liaison Office Building 602 Fort Rucker, AL 36362

French Army Liaison Office USAAVNC (Building 602) Fort Rucker, AL 36362-5021

Australian Army Liaison Office Building 602 Fort Rucker, AL 36362

Dr. Garrison Rapmund 6 Burning Tree Court Bethesda, MD 20817

Commandant, Royal Air Force Institute of Aviation Medicine Farnborough, Hampshire GU14 6SZ UK Defense Technical Information Cameron Station, Building 5 Alexandra, VA 22304-6145

Commander, U.S. Army Foreign Science and Technology Center AIFRTA (Davis) 220 7th Street, NE Charlottesville, VA 22901-5396

Commander Applied Technology Laboratory USARTL-ATCOM ATTN: Library, Building 401 Fort Eustis, VA 23604

Commander, U.S. Air Force
Development Test Center
101 West D Avenue, Suite 117
Eglin Air Force Base, FL 32542-5495

Aviation Medicine Clinic TMC #22, SAAF Fort Bragg, NC 28305

Dr. H. Dix Christensen Bio-Medical Science Building, Room 753 Post Office Box 26901 Oklahoma City, OK 73190

Commander, U.S. Army Missile
Command
Redstone Scientific Information Center
ATTN: AMSMI-RD-CS-R
/ILL Documents
Redstone Arsenal, AL 35898

Aerospace Medicine Team HQ ACC/SGST3 162 Dodd Boulevard, Suite 100 Langley Air Force Base, VA 23665-1995 U.S. Army Research and Technology Laboratories (AVSCOM) Propulsion Laboratory MS 302-2 NASA Lewis Research Center Cleveland, OH 44135

Commander
USAMRMC
ATTN: SGRD-ZC (COL John F. Glenn)
Fort Detrick, Frederick, MD 21702-5012

Dr. Eugene S. Channing 166 Baughman's Lane Frederick, MD 21702-4083

U.S. Army Medical Department and School USAMRDALC Liaison ATTN: HSMC-FR Fort Sam Houston, TX 78234

NVESD AMSEL-RD-NV-ASID-PST (Attn: Trang Bui) 10221 Burbeck Road Fort Belvior, VA 22060-5806

CA Av Med HQ DAAC Middle Wallop Stockbridge, Hants S020 8DY UK

Dr. Christine Schlichting Behavioral Sciences Department Box 900, NAVUBASE NLON Groton, CT 06349-5900

Commander Aviation Applied Technology Directorate ATTN: AMSAT-R-TV Fort Eustis, VA 23604-5577 COL Yehezkel G. Caine, MD Surgeon General, Israel Air Force Aeromedical Center Library P. O. Box 02166 I.D.F. Israel

HQ ACC/DOHP 205 Dodd Boulevard, Suite 101 Langley Air Force Base, VA 23665-2789

41st Rescue Squadron 41st RQS/SG 940 Range Road Patrick Air Force Base, FL 32925-5001

48th Rescue Squadron 48th RQS/SG 801 Dezonia Road Holloman Air Force Base, NM 88330-7715

HQ, AFOMA ATTN: SGPA (Aerospace Medicine) Bolling Air Force Base, Washington, DC 20332-6128

ARNG Readiness Center ATTN: NGB-AVN-OP Arlington Hall Station 111 South George Mason Drive Arlington, VA 22204-1382

35th Fighter Wing 35th FW/SG PSC 1013 APO AE 09725-2055

66th Rescue Squadron 66th RQS/SG 4345 Tyndall Avenue Nellis Air Force Base, NV 89191-6076 71st Rescue Squadron 71st RQS/SG 1139 Redstone Road Patrick Air Force Base, FL 32925-5000

Director
Aviation Research, Development
and Engineering Center
ATTN: AMSAT-R-Z
4300 Goodfellow Boulevard
St. Louis, MO 63120-1798

Commander
USAMRMC
ATTN: SGRD-ZB (COL C. Fred Tyner)
Fort Detrick, Frederick, MD 21702-5012

Commandant
U.S. Army Command and General Staff
College
ATTN: ATZL-SWS-L
Fort Levenworth, KS 66027-6900

ARNG Readiness Center ATTN: NGB-AVN-OP Arlington Hall Station 111 South George Mason Drive Arlington, VA 22204-1382

Director Army Personnel Research Establishment Farnborough, Hants GU14 6SZ UK

Dr. A. Kornfield 895 Head Street San Francisco, CA 94132-2813

ARNG Readiness Center AATN: NGB-AVN-OP Arlington Hall Station 111 South George Mason Drive Arlington, VA 22204-1382 Cdr, PERSCOM ATTN: TAPC-PLA 200 Stovall Street, Rm 3N25 Alexandria, VA 22332-0413 HQ, AFOMA ATTN; SGPA (Aerospace Medicine) Bolling Air Force Base, Washington, DC 20332-6188